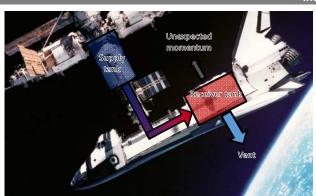
Numerical Model of No-Vent-Fill (NVF)

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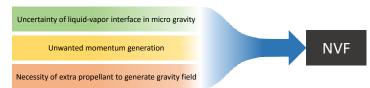
Introduction of NVF



Definition

> No-Vent-Fill: Cryogenic propellant transfer and filling under micro gravity without venting process

Problems of Vent-Fill



Problem & Objective

Objective: Development of a reliable numerical model for NVF

> Developing an analytic model by using previous experimental data

Considering of flash evaporation process at the beginning of the NVF

Advantages

- > Reduction of the spacecraft weight at launching
- Increase of the available fuel to explore space

Problem

System modeling

Assumptions

- Condensation at gas-droplet interface only
- Negligible heat transfer at vaporwall interface
- Uniform temperature of liquid
- Empirical heat transfer coefficient from previous experimental data

Control volumes

- Vapor control volume
- > Liquid control volume
- Liquid-vapor interface control

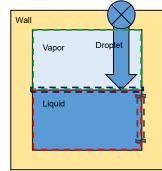
yes

yes

Liquid-wall interface control volume

Process logics

Condensation



no

Result and discussion

Difficulty of predicting the experimental results

■ Propellant selection: R14 (CF₄)

> Reducing experiment costs

- Non flammable
- Similar structure with methane: CH₄

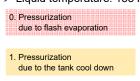
Initial conditions

- > Wall temperature: 180 K
- Supply tank pressure: 400 kPa

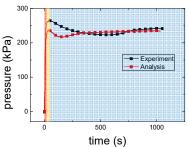
Boundary conditions

> Average mass flow rate: 7 g/s (by using valve coefficient C_v in analysis)

> Liquid temperature: 155 K

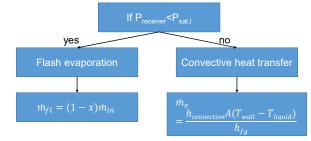


2. Pressure decrease due to condensation

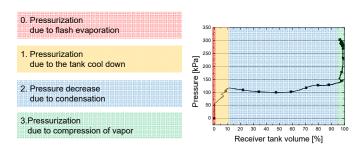


80 iquid level (%) - Analysis 500 1000 time (s)

Evaporation



Processes



Discussions

- The numerical model quite well predicts the real situation; especially for the pressure tendency, the process time and the liquid level.
- Compensation of liquid level by using mass balance correlation due to uncertainty of the level meter.

Objectives

Conclusions

- The first numerical model of NVF with considering the flash evaporation process is constructed.
- More accurate heat transfer correlations are needed.